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The First Two Rows Of This Excel File Help Keep The

Data Organized, But They Are Not Required. 13. In The First Line Of Data, Change The X Coordinate To "1" And The Y Coordinate To "0". The First Data Point Should Be Located At The Trailing Edge Of The Airfoil, But Due To Resolution And Accuracy Limits It Is Not Exactly (1,0). 14. Enter "1" In Cell A3 And A4, And Fill Through The ... Feb 7th, 2024

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#### Aerodynamic Characteristics Of A NACA 4412

#### Airfoil

NACA 4412 Airfoil 4 Digit Code Used To Describe Airfoil Shapes 1st Digit - Maximum Camber In Percent Chord 2nd Digit - Location Of Maximum Camber Along Chord Line (from Leading Edge) In Tenths Of Chord 3rd And 4th Digits - Maximum Thickness In Percent Chord NACA 4412 With A Chord Of 6" Max Camber: 0.24" (4% X 6") Location Of Max Camber: 2.4" Aft Of Leading Edge (0.4 X 6") Feb 8th, 2024

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